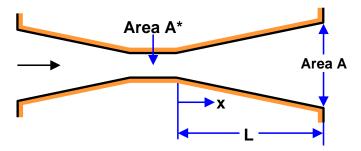
An Internet Book on Fluid Dynamics

Problem 332B

A crude de Laval nozzle with a throat area, A^* , and a diffuser exit area 16 times larger ($A = 16A^*$) is made using a straight-sided conical diffuser as indicated below:



The nozzle is supplied from an air reservoir ($\gamma = 1.4$) of pressure, p_0 ; the external pressure of the air downstream of the diffuser exit is p_E . Find the ratio p_E/p_0 at which a normal shock will form half-way along the diffuser, that is to say at x/L = 0.5.